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rotor failures occurred during the takeoff and climb stages of flight.

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NAVAL AIR PROPULSION CENTER TRENTON, NEW JERSEY 08628

PROPULSION TECHNOLOGY AND PROJECT ENGINEERING DEPARTMENT

NAPC-PE-9 NASA-CR-159474 JULY 1978

ROTOR FRAGMENT PROTECTION PROGRAM: STATISTICS ON AIRCRAFT GAS TURBINE ENGINE ROTOR FAILURES THAT OCCURRED IN U. S. COMMERCIAL AVIATION DURING 1976

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AUTHORIZATION: NASA DPR C-41581-B, MOD.8

ACKNOWLEDGEMENTS

We thank the Flight Standards National Field Office, Federal Aviation Administration, Oklahoma City, Oklahoma, for their cooperative effort in providing the basic data used for this report.

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INTRODUCTION

This report has been prepared as part of the Rotor Fragment Protection Program (RFPP), which is sponsored by the National Aeronautics and Space Administration (NASA)¹ and conducted by the Naval Air Propulsion Center (NAPC). The objective of the RFPP is to develop criteria for the design of devices that will be used on aircraft to protect occupants and the aircraft structure from the potentially lethal and devastating fragments that are generated by gas turbine engine rotor bursts.

Presented in this report are statistics on gas turbine rotor failures that have occurred in U. S. commercial aviation during 1976. These statistics are based on data compiled from the Flight Standards Service Difficulty Reports (SDRs) that were published by the Dapartment of Transportation, Federal Aviation Administration (FAA). The compiled data were analyzed to establish:

- 1. The incidence of rotor failures and the incidence of contained and uncontained 2 rotor fragments.
- 2. The distribution of rotor failures with respect to engine rotor component; i.e., fan, compressor or turbine rotors and their rotating attachments or appendages such as spacers and seals.
- The type of rotor fragment (disk, rim or blade) typically generated at failure.
 - 4. The cause of failure.
 - 5. The type of engines involved.
 - 6. The flight condition at the time of failure.

RESULTS

- 1. The data used for analysis are contained in APPENDIX A. The results of these analyses are shown in Figures 1 through 6.
- a. Figure 1 shows that 186 rotor failures occurred in 1976. These rotor failures accounted for approximately 9.3% of the 2002 shutdowns experienced by the gas turbine powered U. S. commercial aircraft fleet during 1976. Rotor fragments were generated in 109

¹NASA DPR C-41581-B, Mod. 8.

²An uncontained rotor failure is defined as a rotor failure that produces fragments which penetrate and escape the confines of the engine casing.

of the failures experienced and, of these, 20 (18.3% of the fragment producing failures) were uncontained. This represents an uncontained failure rate of 3.2 per million gas turbine engine powered aircraft flight hours, or 1.1 per million engine operation hours. Approximately 6.2 million and 18.9 million aircraft flight and engine operating hours, respectively, were logged by the U. S. commercial aviation fleet in 1976. Because of the potentially catastrophic consequence of the fragment producing failures, these rates are considered to be significantly high.

- b. Figure 2 shows the distribution of rotor failures that produced fragments according to the engine component involved fan, compressor, turbine; the types of fragments that were generated; and the percentage of uncontained failures according to the type fragment generated. These data indicate that:
- (1) The incidence of turbine rotor fragment producing failures was three times greater than that of compressor rotor fragment producing failures; these corresponded to 68.8% and 22.9%, respectively, of the total number of rotor bursts. Fan rotor failures accounted for 8.3% of the uncontained failures experienced.
- (2) Blade fragments were generated in 93.6% of the rotor failures; 11.9% of these were uncontained. The remaining rotor fragment failures (6.4%) produced disk, rim and seal fragments, of which 100% were uncontained.
- c. Figure 3 shows the rotor failure distribution among the types of engines that were affected, and the total number of that type engine in use. It appears that the more recently introduced, larger turbofan engines have experienced the highest rate of rotor failure.
- d. Figure 4 shows what caused the rotor failures to occur. Of the known causes of failure*, the dominant causal factors were: (1) Secondary Causes (45.2%); (2) Foreign Object Damage (36.5%); and (3) Design and Life Prediction Problems (14.8%).
- e. Figure 5 shows the flight conditions that existed when the various rotor failures or bursts occurred. Approximately 67% of the 186 rotor failures occurred during the takeoff and climb stages of flight. Approximately 70% of the rotor fragment producing failures, and 70% of the uncontained rotor failures, occurred during these same stages of flight. The highest percentage of uncontained rotor bursts (45%) were experienced during takeoff.
- f. Figure 6 shows the annual incidence of uncontained rotor failure in commercial aviation for the years 1962 through 1976. During 1976, the incidence of uncontained rotor failure increased over the previous

^{*}Because of the high percentage of unknown causes of rotor failure, the analysis was based on the total number of known causes.

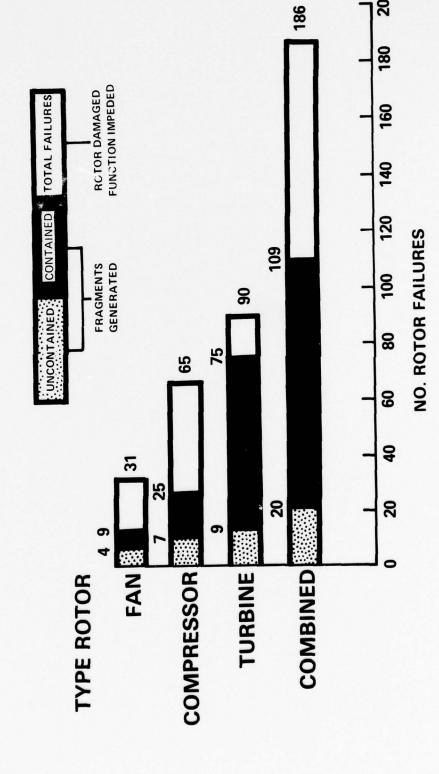
years, 1974 and 1975, by 15 and 30%, respectively. Over the past four years, 1973 thru 1976, an average of 19 uncontained rotor failures per year have occurred. During this same time period, the rate of uncontained rotor bursts has remained relatively constant at an average of approximately 1 per million engine operating hours.

CONCLUSIONS

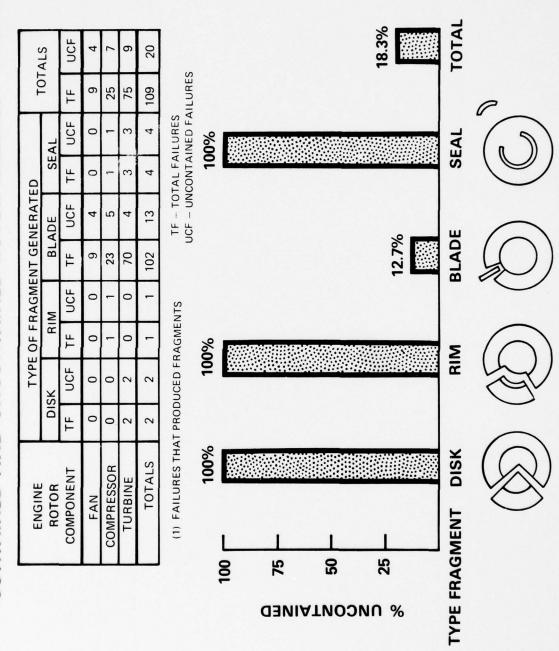
- 1. The incidence of rotor failure and uncontained failure is significantly high enough to warrant continuation of the experimental and analytical efforts that constitute the Rotor Fragment Protection Program.
- 2. Of all the types of fragments generated at rotor failure, disk and fan blade fragments, because of their size, high energy content and high rate of uncontainment, continue to be the threat that must be addressed in the RFPP.
- 3. The number of uncontained blade failures that occurred during 1976 is surprisingly high considering that, under FAA regulations, rotor blade containment is required for engine certification.
- 4. It appears that causes beyond the control or scope of present technology such as FOD, structural life and integrity prediction, and secondary effects, are still primarily responsible for most of the rotor failures that occur.

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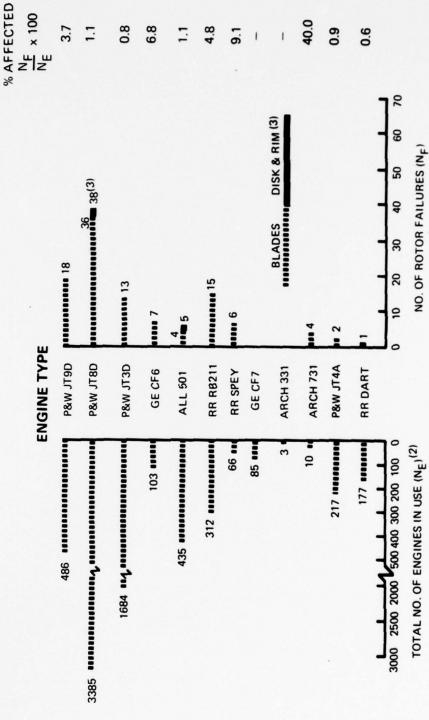
IN U.S. COMMERCIAL AVIATION 1976 INCIDENCE OF ROTOR FAILURE FIGURE 1



CONTAINED AND UNCONTAINED ROTOR FAILURES (1) - 1976 COMPONENT AND FRAGMENT TYPE DISTRIBUTIONS FOR FIGURE 2



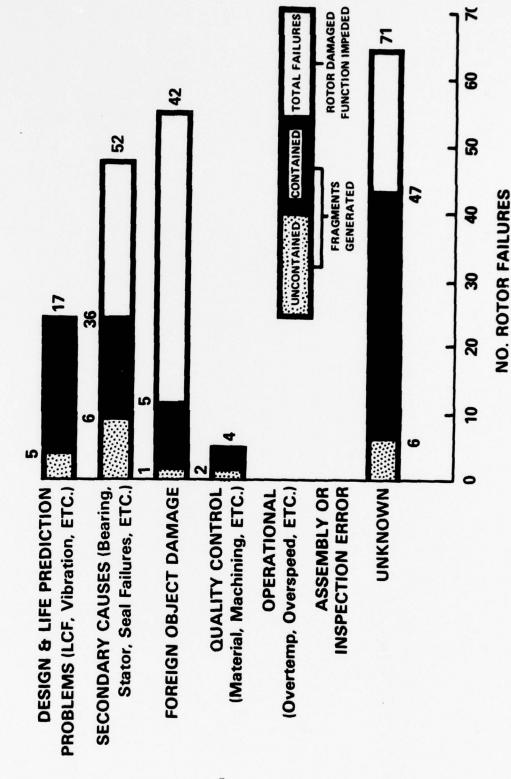
THE INCIDENCE OF ROTOR FAILURE(1) IN .U U.S. COMMERCIAL AVIATION ACCORDING TO ENGINE TYPE AFFECTED - 1976 FIGURE 3



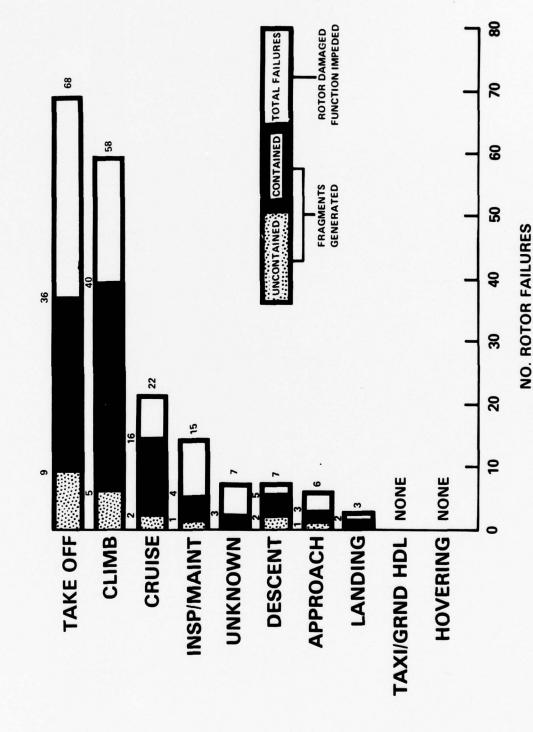
(1) FAILURES THAT PRODUCED FRAGMENTS

(2) YEARLY AVG. OF AIRCRAFT IN USE AT END OF EACH MONTH (3) 1 SEAL FAILURE INCLUDED IN DISK/RIM COMPILATION

ROTOR FAILURE CAUSE CATEGORIES - 1976 FIGURE 4

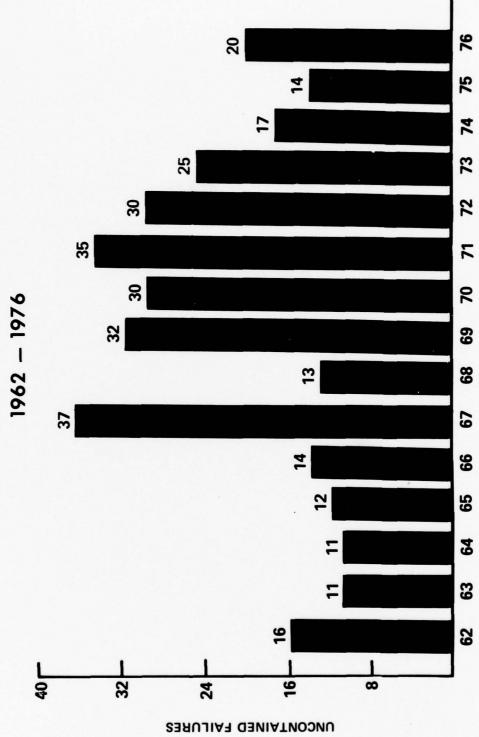


FLIGHT CONDITION AT ROTOR FAILURE - 1976 FIGURE 5



CALENDAR YEAR

THE INCIDENCE OF UNCONTAINED ROTOR FAILURES IN U.S. COMMERCIAL AVIATION FIGURE 6



9

APPENDIX A

Data on Rotor Failures in U. S. Commercial Aviation for 1976. Compiled from the Federal Aviation Administration Service Difficulty Reports.

DATA COMPILATION KEY:

Component Code:

- F Fan
- C Compressor
- T Turbine

Fragment Type Code:

- D Disk adiagraphy valuablible noise agreement about
- R Rim
- B Blade
- S Seal
- N None

Cause Code:

- 1 Design and Life Prediction Problems
- 2 Secondary Causes
- 3 Foreign Object Damage
- 4 Quality Control
- 5 Operational
- 6 Assembly and Inspection Error
- 7 Unknown

Containment Condition Code:

- C Contained
- NC Not Contained
- N No Fragments Generated

Flight Condition Code:

- 1 Insp/Maint
- 2 Taxi/Grnd Hdl
- 3 Takeoff
- 4 Climb
- 5 Cruise
- 6 Descent
- 7 Approach
- 8 Landing
- 9 Hovering
- 10 Unknown

CHARACTERISTICS OF ROTOR FAILURES - 1976

						FRAGMENT		CONTAINMENT	FLICHT
SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	TYPE	CAUSE	CONDITION	CONDITION
01136030	1/13	TXI	6-00	JT8D	H	æ	1	υ	7
01136032	1/4	TWA	L-1011	RB211	H	æ	7	· 0	4
01266024	1/13	VAL	DC-10	CF6	H	В	1	NC	5
01266066	1/26	EAL	L-1011	RB211	H	В	7	O	4
01266026	1/26	WAL	UNDET.	JT8D	υ	S	7	NC	-
01266049	1/26	EAL	L-1011	RB211	H	В	7	v	4
02036029	1/18	AAA	BA111	SPEY	T.	Д	7	O	3
02056031	1/27	TWA	B-747	JT9D	F	ø	2	v	3
02096030	1/28	SAAX	L-382	501	v	Ø	2	v	2
02096032	1/26	NWA	DC-10	JT9D	ţz.	æ.	1	NC	2
02116039	2/1	NCA	6-00	JT8D	F	æ	2	311 01	9
02186026	2/1	AWI	6-0d	JT8D	mac H	B	7		
02196035	2/8	TWA	B-747	JT9D	eroli erali	В	2	Taxi Taxi Taxi O	4
02266068	2/26	EAL	L-1011	RB211	- 0 O	8	2	υ	7 4 di
02256036	5/6	PAA	B-747	JT9D	Įz.	æ	4	NC	14
03016033	2/19	UAL	B-727	JT8D	H	æ	2	NC	3

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

	SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
8	03036030	2/22	TWA	B-747	JT9D	T	В	7	υ	7
	03036031	2/20	TWA	B-727	JT8D	T	В	2	υ	4
	03046029	2/22	SAAX	L-382	501	T	В	7	o	4
	03056031	2/25	NCA	6-0Q	JT8D	T	æ	7	υ	4
	03236020	3/9	FAL	B-737	JT8A	ပ	, M	3	v	5
	03226024	3/4	TWA	B-727	JT8D	Т	æ	2	v	7
	03116039	3/11	EAL	L-1011	RB211	П	æ	7	v	4
	04126035	3/28	NAL	DC-10	CF6	ĵz,	æ	4	υ	3
	04126036	3/28	NAL	B-727	JT8D	T	g	2	v	3
	04226029	4/12	TWA	B-707	JT3D	T	æ	7	υ	7
	04016028	3/15	AAA	6-0Q	JT8D	ပ	В	7	v	3
	04296037	4/19	TWA	B-747	JT9D	T	S	2	NC	3
	04306036	3/31	DAL	DC-10	CF6	T	æ	1	o	3
	04306037	4/12	TSA	B-737	JT8D	T	æ	1	O	3
	05056123	5/5	DAL	L-1011	RB211	1	æ	2	v	10
	05066031	4/20	NWA	DC-10	JT9D	[tu	æ	60	O	1

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
05056121	5/5	EAL	L-1011	RB211	T	В	2	S	3
05066029	4/27	PAA	B-707	JT3D	υ	В	2	NC	9
05196017	6/9	ONAS	DC-8	JT3D	v	В	2	S	7
05196018	2/1	PAI	B-737	JT8D	Т	В	7	υ	7
04276027	4/18	AAL	B-747	JT9D	T	В	3	S	4
05146080	5/14	EAL	L-1011	RB211	T	В	7	C	2
05256037	4/27	PSAX	UNDET.	JT8D	Т	В	2	O	1
05276042	5/18	AAL	B-747	JT9D	T	S	2	NC	3
06046034	5/13	BCAT	LR-35	TFE731	T	В	1	v	9
06026036	5/19	AAA	6-DQ	JT8D	T	В	1	O	3
06016087	6/1	EAL	L-1011	RB211	H	Д	2	v	4
06076035	5/21	AAA	BA111	SPEY	Т	В	7	v	3
06106024	6/1	TWA	B-747	JT9D	T	В	2	S	7
06116102	6/11	EAL	L-1011	RB211	Т	В	7	v	10
06146023	6/2	TWA	B-727	JT8D	T	æ	7	S	4
06226024	6/14	ACAX	B-737	JT8D	H	8	7	o o	3
06226025	6/9	OZA	0C-9	JT8D	H	В	1	O	8

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

						1			
SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
06306030 8 6/21	6/21	TWA	B-727	JT80	1	æ	7	v	8
07076031	97/9	TWA	1-1011	RB211	1	æ	1	υ	2
07226032	7/14	AAL	B-707	JT3D	(Es	æ	4	NC	8
07286033	7/18	NAL	B-727	JT8D	ပ	æ	7	NC	8
07306032	1/21	AAL	B-727	JT8D	ĵ2 ₄	æ	7	υ	7
08036030	1/21	0ZA	6-DQ	JT8D	H	æ	2	v	7
08036031	7/19	NWA	B-747	JT9D	υ	ø	2	_o	7
08056032	8/1	AAL	DC-10	CF6	υ	æ	7	NC	7
60096080	7/22	DAL	DC-8	JT3D	<u>Fu</u>	æ	4	O	∞
08295030	8/29	FAL	CV-580	501	H	Q		NC	7
08265034	8/26	TWA	B-747	JT9D	T	S	7	NC	4
08126032	7/29	NWA	B-747	JT9D	н	æ	7	o	æ
08136028	7/28	NWA	DC-10	JT9D	H.	æ	-	S	3
08186027	8/8	TWA	B-747	JT9D	н	EQ	7	S	3
08266033	8/17	AAL	B-727	JT8D	<u>p</u> s.	æ	2	S	3
08206032	6/8	AWI	09	JT8D	н	B	7	S	4
08236032	8/12	PCTC	066-AD	CJ805	<u>Bu</u>	æ	3	NC	3

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
08256030	8/13	nos	6DC	JT8D	H	89	7	υ	3
08276032	8/13	FTLS	9-20	JT3D	၁	. ≃	1	NC	3
08276035	8/18	NCA	6-0d	JT8D	L	В	7	υ	3
09026028	9/8	BCAT	LR-35	TFE731	ı	В	1	υ	2
09096029	8/29	NAL	DC-10	CF6	T.	89	7	υ	5
09096031	8/30	TWA	B-747	JT9D	Ţ	В	2	υ	2
09106036	8/31	ACAX	B-737	JT8D	1	g	7	υ	3
09106037	9/1	TWA	L-1011	RB211	T	В	1	υ	4
09176028	8/27	CAPS	DC-8	JT3D	T	В	7	ပ	3
09206030	8/31	CAPS	DC-8	JT4A	T	В	7	ນ	2
09286024	9/21	TWA	B-707	JT3D	T	æ	7	v	4
09286026	9/17	WZ0	6-0Q	JT8D	υ	В	2	ပ	2
09296020	1/6	DAL	B-727	JT8D	υ	В	2	v	2
10016019	9/19	AWI	6-DQ	JT8D	н	В	1	v	10
10056025	9/21	AAA	BA111	SPEY	ь	æ	7	O	4
10066025	9/3	PSAX	B-727	JT8D	Т	В	7	o	2
10046023	10/4	PAA	B-747	JT9D	H	Д	2	U	25

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

ON BUS	TATE	GITHMITTED	ATDCDART	FNCTNE	TWEND	FRAGMENT	CATICE	CONTAINMENT	FLIGHT
SUN MO	DATE	SOBILLIEN	ATHOMALI	ENGINE	COUR ONEINT	TILE	CAUSE	CONDITION	CONDITION
10156025	10/1	CAL	DC-10	CF6	1	æ	7	O	7
10196019	10/7	TWA	B-707	JT3D	υ	æ	7	S	3
10196021	10/9	TWA	B-727	JT8D	ပ	В	7	S	7
10206024	10/11	AAA	BA111	SPEY	_O	В	7	S	3
10196019	10/7	TWA	B-707	JT3D	ပ	В	7	C	7
10305019	10/30	KTS	6-DC	JT8D	T	В	7	NC	3
10196021	10/9	TWA	B-727	JT8D	v	В	7	S	7
10216025	10/12	AAA	BA111	SPEY	ပ	В	7	S	3
10226026	10/15	TWA	B-707	JT4A	Т	В	2	S	4
10276016	10/5	BCAT	LR-35	TFE731	н	В	-	ပ	7
11176023	11/4	WAL	B-720	JT8D	v	В	7	NC	3
11036027	10/20	FAL	CV-580	501	v	В	7	O	4
11096021	10/21	AAA	BA111	909	Т	В	7	O	3
11105026	1/11	WAA	F-27B	DART	T	D	1	NC	9
11236020	1/12	TWA	B-747	JT9D	v	В	2	v	4
11266032	11/13	EAL	L-1011	RB211	T	æ	1	c	5
12076025	11/29	TWA	L-1011	RB211	v	В	2	S	4

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
12096023	11/30 TWA	TWA	B-727	JT8D	v	æ	7	NC	4
12146024	11/11	EAL	B-727	JT8D	υ	29	2	O	4
12166025	12/8	TWA	B-707	JT3D	Ħ	82	7	S	3
12296026	11/10 BCAT	BCAT	LR-35	TFE731	H	23	9	v	5
05106030	5/10	UAL	DC-10	CF6	T	В	7	٥	1
01107023	12/27 FAL	FAL	CV-580	501	I	В	7	S	80
01127031	12/30 BNF	BNF	BC-8	JT3D	H	В	7	υ	7
01177032	12/23 UAL	UAL	B-727	JT8D	Н	В	7	NC	7
02017037	1/22	EAL	L-1011	RB211	T	В	7	υ	3
04266036	4/26	NCA	CV-580	501	Ų	Z	ന	Z	3
12016024	12/1	SAAS	1886	501	v	Z	6	Z	3
05076030	2/1	FECT	FALCON	CF7	S	Z	3	Z	4
10276018	10/27 MIAS	MIAS	DC-8	JT4A	O	N	3	Z	8
04136037	3/25	AAA	93d	JT8D	S	Z	3	N	3
06116016	5/30	AAIX	B-720	лтзс	O	Z	3	Z	1
06106025	5/31	PAA	B-747	JT9D	Œ	Z	3	Z	7
05106028	4/29	ACAX	L-188C	501	v	Z	e -	N	4

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
02026023	2/2	SAAS	L-382	501	υ	z	3	Z	7
05116028	4/19	FECT	FALCON	CF7	υ	z	3	Z	2
03056030	2/22	PAA	B-707	JT3D	υ	z	3	Z	3
08106030	8/10	DAL	DC-8	JT3D	ĵz,	Z	9	Z	3
12016026	11/21 TWA	TWA	B-707	JT3D	ĵz,	Z	6	z	3
04266037	4/26	PAA	B-707	JT3D	ĵz,	=	3	Z	2
07216030	6/1	TIAS	DC-8	JT3D	υ	N	8	N	4
10196020	10/7	ONAS	DC-8	JT4	υ	z	3	N	9
04146031	3/30	AAA	6-DQ	JT8D	<u>P</u>	z	3	N	3
10296013	10/26 SWAX	SWAX	B-737	JT8D	(E)	z	3	N	1
02266068	2/26	EAL	L-1011	RB-211	υ	z	3	N	7
08266032	8/8	PCTC	066-AD	CJ805	Œ4	z	3	N	1
01166027	1/5	AAA	0C-9	JT8D	v	z	7	Z	4
01196031	1/1	AAA	00a	JT8D	ı	z	1	Z	3
04196034	3/22	PSAX	UNDET.	JT8D	ь	z	7	N	10
04236036	3/6	FECT	MD-20	CF7	v	N	7	N	4
04236038	3/22	FECT	MD-20	CF7	(Es	z	7	z	1

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

FLIGHT	5	2	3	3	3	3	3	3	10	1	10	3	1	3	1	3	7
CONTAINMENT	Z	Z	Z	Z	2.	Z	Z	Z	Z	Z	Z	Z	Z	Z	Z	Z	Z
CAUSE	8	2	3	3	3	3	3	8	3	2	2	3	2	7	7	8	8
FRAGMENT	z	Z	Z	Z	N	Z	Z	Z	Z	Z	Z	Z	Z	z	Z	Z	z
COMPONENT	ĵu,	υ	v	£4	Ē4	Œ.	<u>Eu</u>	T	£4	ı	1	U	£4	υ	v	1	U
ENGINE	JT3D	CF6	JT8D	JT8D	JT8D	JT8D	JT8D	501	JT8D	JT8D	JT8D	501 (JT8D 1	RB-211 (501	SPEY	501 (
AIRCRAFT	B-707	DC-10	6-0q	B-727	09	0-0q	B-727	L-382	6-2Q	6-2Q	UNDET.	L-382	B-737	L-1011	L-382	BA-111	L-382
SUBMITTER	PAA	NAL	AAA	BNF	OZA	OZA	TWA	AIAX	OZA	AAA	PSAX	SAAS	FAL	EAL	SRAX	AAA	SAAX
DATE	4/12	5/2	6/9		6/14	6/14		6/14								1/14	1/17
SDR NO.	04266037	05136028	05206023	06236009 6/10	06246021	06306017	07126008 6/28	0708630	07166014 7/3	07206020 7/5	08136026 6/22	08206030 8/20	08266034 8/6	10186025 10/10	01277030 1/4	02027038	02026023

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

SDR NO.	DATE	SUBMITTER	AIRCRAFT	ENGINE	COMPONENT	FRAGMENT	CAUSE	CONTAINMENT	FLIGHT
05196013	5/5	PCTC	L-188	501	o o	Z	7	z	4
11016021	10/26 NAL	NAL	DC-10	CF6	υ	z	7	z	3
09206028	9/6	NAL	DC-10	CF6	Ŀ	z	7	z	3
03306023	3/17	NAL	DC-10	CF6	v	z	7	z	3
05076030 5/7	2/1	FECT	MD-20	CF7	υ	z	e	Z	4
11176024 11/17 TWA	11/11	TWA	B-707	JT3D	υ	2	6	z	5
02096030 1/28	1/28	PAI	B-737	JT8D	υ	z	7	z	4
08276034 8/15	8/15	TXI	6-DC	JT8D	į×ι	Z	2	Z	3
03106030 3/10	3/10	CAL	B-727	JT8D	(Fa)	z	2	Z	
03306024 3/30	3/30	CAL	B-727	JT8D	<u>p</u>	z	2	Z	1
09096030 8/27	8/27	CAL	B-727	JT8D	v	Z	2	Z	4
07166030 7/3	1/3	TWA	B-727	JT8D	F	z	7	Z	4
12296022	12/15	TWA.	B-727	JT8D	F	z	1	Z	4
11096020	10/19	PSAX	B-727	JT8D	F	z	2	Z	2
07096032	6/1	AAA	6-0q	JT8D	F	z	2	Z	1
09156026	9/5	TWA	B-747	JT9D	[Eq	z	8	Z	3
03246031	3/14	TWA	B-747	JT9D	ĵz,	Z	2	Z	4

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

FLIGHT	4	3	4	3	3	3	3	9	3	8	4	3	1	4	3	10
CONTAINMENT	z	Z	z	Z	Z	N	z	Z	Z	Z	Z	Z	N	Z	×	Z
CAUSE	7	7	7	7	7	7	7	7	7	3	2	7	2	7	3	2
FRAGMENT	z	Z	Z	z	z	N	Z	z	z	z	Z	Z	Z	Z	Z	Z
COMPONENT	υ	v	υ	υ	υ	υ	U	v	ပ	ĵs.	н	T	H	υ	υ	н
ENGINE	JT9D	RB211	RB211	RB211	RB211	RB211	RB211	RB211	RB211	RB211	RB211	RB211	RB211	JT8D	501	501
AIRCRAFT	B-747	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	L-1011	B-737	CV-580	L-188
SUBMITTER	FTLS	EAL	EAL	EAL	EAL	EAL	EAL	EAL	EAL	DAL	EAL	EAL	EAL	PAI	NCA	SAAS
DATE	3/11	3/31	4/28	3/17	4/19	6/1	3/5	1/16	6/4	2/6	2/25	1/16	6/24	1/22	9/15	9/22
SDR NO.	03116031	0316138	04286087	03176073	04196046	06091090	03056069	01166060 1/16	04096102	02066074	02256066	01166063	06246052	02046033	08196030	10016018

CHARACTERISTICS OF ROTOR FAILURES - 1976 (Continued)

FLIGHT		
17 S	3	7
CONTAINMENT	Z	Z
CAUSE	3	7
FRAGMENT	z	Z
COMPONENT	v	£.
ENGINE	лгэс	RB211
AIRCRAFT	B-720	L-1011
SUBMITTER	MIAS	TWA
DATE	12/26	9/1
SDR NO.	01107025	09106037

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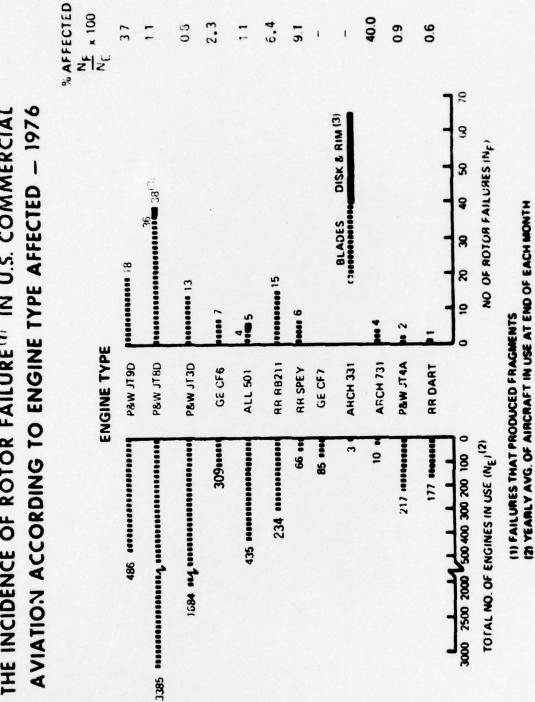
Encl: (1) NAPC Figure 3 The Incidence Of Rotor Failure (1) In U.S. Commercial Aviation According To Engine Type

Affected - 1976 - Page 6 (corrected copy)

1. Enclosure (1) is being forwarded for replacement of page 6 in subject report.

) J. L. MOSS By direction

THE INCIDENCE OF ROTOR FAILURE (*) IN U.S. COMMERCIAL FIGURE 3



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(3) 1 SEAL FAILURE INCLUDED IN DISK/RIM COMPILATION